

GOVERNMENT OF INDIA OFFICE OF THE DIRECTOR GENERAL OF CIVIL AVIATION TECHNICAL CENTRE, OPPOSITE SAFDRJUNG AIRPORT, NEW DELHI

CIVIL AVIATION REQUIREMENTS
SECTION 6 – Design Standards & Type Certification
SERIES 'A', PART I
16th December, 2014

Subject: Requirements for validation of Type Certificate of aeronautical products (Aircraft, engine and propeller)

1. Introduction:

Rule 49B of 'the Aircraft Rules,1937' empowers Director General of Civil Aviation (DGCA) to validate Type Certificates (TC) for aeronautical products, designed/manufactured in another country and type certificated by the regulatory authority of that country. This part of the Civil Aviation Requirements lays down the procedure and requirements for validation of Type Certificate, and also lays down the roles and responsibilities of the Validating Authority (VA) and Certificating Authority (CA) for validation activities and post-validation activities.

This CAR is issued under the provisions of Rule 133A of 'the Aircraft Rules, 1937'.

2. Definitions and acronyms:

- a) Aeronautical product means any civil aircraft, aircraft engine and propeller
- b) BASA-IPA means Bilateral Aviation Safety Agreement-Implementation Procedures for Airworthiness & available on DGCA website www.dgca.nic.in.
- c) CA means Certificating Authority, Civil Aviation Authority of issuance of Type certificate.
- d) VA means Validating Authority i.e. DGCA, India.

3. Applicability

3.1 DGCA shall validate the Type Certificate of aeronautical product, provided:

EFFECTIVE: FORTHWITH

- (a) the aeronautical product is type certificated by a foreign Civil Aviation Authority (CAA) and the Indian Organisation intends to manufacture the aeronautical product in India under license agreement from foreign OEM/Type Certificate Holder; or
- (b) the aeronautical product is proposed to be imported by Indian operator, and is type certificated by a CAA whose certification specification /design code and procedures are not in harmony with FAA/EASA.

4. Requirements

- 4.1 Type validation shall require Bilateral Aviation Safety Agreement-Implementation Procedures for Airworthiness (BASA-IPA) or Working Arrangement or Memorandum of Understanding (MOU) between DGCA and CAA of the State of Design. Such arrangements/ agreements shall define roles and responsibilities of VA and CA including post-type validation activities.
- 4.2 The application shall be forwarded through the Civil Aviation Authority of the State of Design to DGCA.
- 4.3 The applicant shall furnish the requisite information in the following Annexures to this CAR, to the Director, Aircraft Engineering Directorate (AED), DGCA, Opposite Safdarjung Airport, New Delhi 110003, along with requisite fees as per Rule 62 of the Aircraft Rules, 1937 in the form of a demand draft payable to PAO, DGCA, MCA, New Delhi.
 - a) Annexure I For validation of Type Certificate of Airplane
 - b) Annexure II For validation of Type Certificate of Rotorcraft
 - c) Annexure III For validation of Type Certificate of Engine
 - d) Annexure IV For validation of Type Certificate of Propeller
 - Note: 1. The certificates and documents issued by the CAA should be in English language only.
 - 2. Type Validation of aeronautical product by AED of DGCA shall be from design view point only.
- 4.4 DGCA at its discretion may ask any additional information/ documentation/ clarification from the TC holder.
- 4.5 Special conditions may be imposed on foreign TC by DGCA in specific cases for safe operation of the aeronautical product in India.
- 4.6 The applicant shall arrange familiarization programme of adequate duration for DGCA (AED) officials, depending on the complexity, type of aircraft, type of design change (significant), at their facility. In addition, witnessing of critical tests by DGCA officials may be necessary.

4.7 Service experience change:

4.7.1 Where the DGCA finds that an unsafe condition exists with respect to a design feature or characteristic of the validated aeronautical product, the Director General may issue necessary instruction specifying conditions and limitations including inspections for continued airworthiness or may altogether prohibit the use of the same till the unsafe conditions are corrected.

4.7.2 When design changes are considered necessary, the holder of the TC shall submit appropriate design changes for approval of the DGCA. In addition, the TC holder/CAA of the 'state of design' shall immediately inform the DGCA regarding information related to (a) change in design of the validated product, (b) changes/revisions to all applicable manuals, (c) continued airworthiness related information, (d) any other information deemed necessary from safety point of view.

4.8 Exceptions:

Notwithstanding the above requirements, DGCA may exempt/ waive/ reject any of the requirements spelt in this CAR, provided there is a genuine justification for doing so.

5. Validation of Type Certificate:

If the DGCA is satisfied that the aeronautical product conforms to the relevant acceptable standards in respect of design and performance (with such exceptions as may be permitted), the Director General may validate the TC issued by the CAA of the 'state of design' in respect of the aeronautical product.

6. Cancellation, suspension or endorsement of validated type certificate:

If at any point of time, DGCA finds that there is a reasonable ground to indicate that the safety of the aircraft is imperiled because of an unsafe condition developed in the product, the Director General may cancel, suspend or endorse the validated type certificate or may incorporate any modification till the validated type certificate remains in force.

(Dr.-Prabhat Kumar)
Director General of Civil Aviation

Annexure- I

Government of India Directorate General of Civil Aviation			
Aircraft Engineering Directorate Application for validation of Type Certificate of Airplane			
Name and Full address of the Type Certificate (TC) Holder (including e-mail, telephone and fax).	J GOTTINIGATO C	<u> </u>	
Engineering description of airplane with illustrations			
3. Airplane model			
4. Aircraft Category	Normal	Transport	Others
5. State of Design			
6. State of Manufacture			
Type Certificate number & Year of type certification by state of design (Copy to be enclosed)			
8. TCDS (Copy to be enclosed)			
Type certification in other countries (if yes, enclose copies)			
10. Serial number eligible			
11. Certification basis			
12. Production basis			
13. Dimensions			
a. Wing span			
b. Wing aspect ratio			
c. Length overall			
d. Height overall			
e. Wheel track			
f. Wheel base			
g. Wing area			
14. Minimum crew			
15. Number of seats			
16. Type of engine			
17. Number of engines			
18. Engine limits			
19.ETOPS/EDTO certification basis, if applicable			
20. Propeller & propeller limits			
21.Fuel			
22. Fuel Capacity			
23. Oil			
24. Oil Capacity			

25. Airspeed limit	
 a. Maximum operating speed 	
 b. Maneuvering speed 	
 c. Maximum flap extended speed 	
d. Single engine minimum control	
speed	
e. Landing gear extended speed	
f. Demonstrated cross wind	
component	
26. Datum	
27. Leveling means	
28.C.G. range	
29. Weights in kg	
a. Maximum take-off weight	
b. Maximum landing weight	
c. Zero fuel weight	
d. Maximum taxiing weight	
30. Maximum Baggage	
31. Weight and Balance report	
32. Flight Manual	
33. Structural Repair Manual etc	
34. Control surface movements	
a. Wing flaps,	
b. Ailerons,	
c. Elevator,	
d. Trim tab	
e. Rudder	
35. Maximum operating altitude	
36. Maximum operating temperature	
37. Import requirements	
38. Operating manuals	
39. Airworthiness limitations	
40. All applicable Airworthiness Directives	
41. Service information & documents	
42. Maintenance Review Board (MRB) report	
and MMEL for aircraft type certificated in	
transport category	
43. Equipment	
44. Certification Compliance document/	
Compliance matrix	
45. Flight test reports	
46. Supplemental Type Certificate, if	
applicable	
47. Compliance with special conditions such	
as novel or unusual design features in the	
aircraft and substantiation reports.	

SECTION 6 16TH DECEMBER, 2014

48. Compliance documents related to high intensity radiation field (HIRF) effects.	
49. Operational capability from high altitude (>10000 feet) airports. This requirement is mandatory for pressurized transport	
category aircraft. 50. Capability to operate at high ambient temperatures up to +50°C at sea level. Engine cooling test report during take-off/landing phases of flight at Sea Level should be provided in support of operational capability.	
51. Compliance with Noise standard per ICAO guidelines	
52. Letter of intent from an Indian operator (stating the kind of operation the aircraft will be engaged to).	
53. Substantiation documents providing the ba sis for 'Equivalent Level of Safety Findings' and 'Exemptions'.	
54. Access to web portal for necessary documents for continued airworthiness support (user ID, Password, etc.).	
55. Details of fees paid in accordance with Rule 62 of Aircraft Rules 1937 (Demand Draft No. /Indian Postal Order No./amount/Name of the Bank).	
Declaration:	
I hereby declare that the particulars entered in the true and correct to the best of my knowledge and	
Date:	Signature of the applicant Designation

Note:

SECTION 6 16TH DECEMBER, 2014

ANNEXURE II

Government of India Directorate General of Civil Aviation			
Aircraft Engineering Directorate			
Application for validation of Type Certificate of Rotorcraft			
1. Name and Full address of the Type Certificate (TC) Holder (including e-mail, telephone and fax).			
Engineering description of rotorcraft with illustrations			
3. Rotorcraft model	NI	T	Otherna
4. Aircraft Category	Normal	Transport	Others
State of Design State of Manufacture			
7. Type Certificate number & Year of type certification by state of design (Copy to be enclosed)			
8. TCDS (Copy to be enclosed)			
Type certification in other countries (if yes, enclose copies)			
10. Serial number eligible			
11. Certification basis			
12. Production basis	I		
13. Dimensions a. Diameter of main rotor b. Main rotor blade chord c. Diameter of tail rotor d. Length overall e. Length of fuselage f. Height overall g. Wheel track h. Wheel base			
14. Minimum crew			
15. Number of seats			
16. Type of engine			
17. Number of engines			
18. Engine operating limits a. All engines operating b. One engine inoperative limits 19. Rotor limits	Power off Power on		
20. Fuel 21. Fuel Capacity			
	l .		

SECTION 6 16TH DECEMBER, 2014

22. Oil	
23. Oil Capacity	
24. Airspeed limit	
25. Datum	
26. Leveling means	
27. C.G. range	
28. Maximum take-off weight in kg	
29. Maximum Baggage	
30. Weight and Balance report	
31. Flight Manual	
32. Structural Repair Manual etc	
33. Rotor blades and control movements	
34. Maximum operating altitude	
35. Maximum operating temperature	
36. Import requirements	
37. Operating manuals	
38. Airworthiness limitations	
39. All applicable Airworthiness Directives	
40. Service information & documents	
41. Maintenance Review Board (MRB) report	
and MMEL for aircraft type certificated in	
transport category	
42. Equipment	
43. Certification Compliance document/	
Compliance matrix	
44. Flight test reports	
45. Supplemental Type Certificate, if	
applicable	
46. Compliance with special conditions such	
as novel or unusual design features in the	
aircraft and substantiation reports.	
47. Compliance documents related to high	
intensity radiation field (HIRF) effects.	
48. Operational capability from high altitude	
(>10000 feet) airports. This requirement is	
mandatory for pressurized transport	
category aircraft.	
49. Capability to operate at high ambient	
temperatures up to +50°C at sea level.	
Engine cooling test report during take-off/	
landing phases of flight at Sea Level	
should be provided in support of	
operational capability.	
50. Compliance with Noise standard per ICAO	
guidelines	

SECTION 6 16TH DECEMBER, 2014

51. Letter of intent from an Indian operator	
(stating the kind of operation the aircraft	
will be engaged to).	
52. Substantiation documents providing the ba	
sis for 'Equivalent Level of Safety	
Findings' and 'Exemptions'.	
53. Access to web portal for necessary	
documents for continued airworthiness	
support (user ID, Password, etc.).	
54. Details of fees paid in accordance with	
Rule 62 of Aircraft Rules 1937 (Demand	
Draft No. /Indian Postal Order	
No./amount/Name of the Bank).	
Declaration:	
I hereby declare that the particulars entered in the	nis application and the appendices are
true and correct to the best of my knowledge and	d belief.
Date:	Signature of the applicant
	Designation

Note:

SECTION 6 16TH DECEMBER, 2014

ANNEXURE III

Government of India Directorate General of Civil Aviation Aircraft Engineering Directorate		
Application for validation of Type Certificate of Engine		
1. Name and Full address of the Type Certificate (TC) Holder (including e-mail, telephone and fax).		
Engineering description of engine with illustrations		
3. Engine model		
4. Engine Type	Reciprocating Turbojet	Turboprop
5. State of Design		
State of Manufacture		
7. Type Certificate number & Year of type certification by state of design (Copy to be enclosed)8. TCDS (Copy to be enclosed)		
9. Type certification in other countries (if yes,		
enclose copies)		
10. Serial number eligible		
11. Certification basis		
12. Production basis		
13. General Dimensions		
14. Engine limits		
15. Fuel		
16.Oil		
17. Weights in kg		
18. Import requirements		
19. Operating manuals		
20. Airworthiness limitations		
21. All applicable Airworthiness Directives		
22. Service information & documents		
23. Certification Compliance document/ Compliance matrix		
24.Test reports		
25. Supplemental Type Certificate, if applicable		
26. Compliance with special conditions such as novel or unusual design features in the aircraft and substantiation reports.		

SECTION 6 16TH DECEMBER, 2014

 Compliance with Noise standard per ICAO guidelines 	
28. Compliance with Emission standard per ICAO guidelines	
29. Capability to operate at high ambient temperatures up to +50°C at sea level. Engine cooling test report during take-off/landing phases of flight at Sea Level should be provided in support of operational capability.	
30. Letter of intent from an Indian operator (stating the kind of operation the aircraft will be engaged to).	
31. Substantiation documents providing the ba sis for 'Equivalent Level of Safety Findings' and 'Exemptions'.	
32. Access to web portal for necessary documents for continued airworthiness support (user ID, Password, etc.).	
33. Details of fees paid in accordance with Rule 62 of Aircraft Rules 1937 (Demand Draft No. /Indian Postal Order No./amount/Name of the Bank).	
Declaration:	
I hereby declare that the particulars entered in the true and correct to the best of my knowledge and	• • • • • • • • • • • • • • • • • • • •
Date:	Signature of the applicant Designation

Note:

SECTION 6 16TH DECEMBER, 2014

ANNEXURE IV

Government of India Directorate General of Civil Aviation Aircraft Engineering Directorate		
Application for validation of Type Certificate of Propeller		
1. Name and Full address of the Type Certificate (TC) Holder (including e-mail, telephone and fax).	•	
Engineering description of propeller with illustrations		
Propeller model Propeller Type State of Design		
State of Design State of Manufacture Type Contificate purply on 8 Year of type		
7. Type Certificate number & Year of type certification by state of design (Copy to be enclosed)		
8. TCDS (Copy to be enclosed)9. Type certification in other countries (if yes, enclose copies)		
10. Serial number eligible 11. Certification basis		
12. Production basis 13. General Dimensions		
14. Weights in kg 15. Import requirements		
16. Operating manuals 17. Airworthiness limitations		
18. All applicable Airworthiness Directives		
19. Service information & documents		
20. Certification Compliance document/ Compliance matrix		
21. Test reports 22. Supplemental Type Certificate, if		
applicable 23. Letter of intent from an Indian operator		
(stating the kind of operation the aircraft will be engaged to).		
24. Substantiation documents providing the ba sis for 'Equivalent Level of Safety Findings' and 'Exemptions'.		

SECTION 6 16TH DECEMBER, 2014

25. Access to web portal for necessary	
documents for continued airworthiness	
support (user ID, Password, etc.).	
26. Details of fees paid in accordance with	
Rule 62 of Aircraft Rules 1937 (Demand	
Draft No. /Indian Postal Order	
No./amount/Name of the Bank).	
Declaration:	
I hereby declare that the particulars entered in the	nis application and the appendices are
true and correct to the best of my knowledge and	d belief.
, ,	
Date:	Signature of the applicant
	Designation
	Ğ

Note: